

A nonlinear beam model to describe the postbuckling of wide neo-Hookean beams

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Abstract

Wide beams can exhibit subcritical buckling, i.e. the slope of the force-displacement curve can become negative in the postbuckling regime. In this paper, we capture this intriguing behaviour by constructing a 1D nonlinear beam model, where the central ingredient is the nonlinearity in the stress-strain relation of the beam's constitutive material. First, we present experimental and numerical evidence of a transition to subcritical buckling for wide neo-Hookean hyperelastic beams, when their width-to-length ratio exceeds a critical value of 12%. Second, we construct an effective 1D energy density by combining the Mindlin-Reissner kinematics with a nonlinearity in the stress-strain relation. Finally, we establish and solve the governing beam equations to analytically determine the slope of the force-displacement curve in the postbuckling regime. We find, without any adjustable parameters, excellent agreement between the 1D theory, experiments and simulations. Our work extends the understanding of the postbuckling of structures made of wide elastic beams and opens up avenues for the reverse-engineering of instabilities in soft and metamaterials.

Keywords: Postbuckling, Nonlinear elastic materials, Wide beam theory

1. Introduction

Recent years have seen an upsurge of interest in the instabilities and postinstability behaviour of flexible structures. Rather than seeing instabilities as failure, they recently have been leveraged to achieve novel functional (meta)materials and structures (Reis, 2015; Reis et al., 2015). As such, materials and structures featuring snapping (Holmes and Crosby, 2007; Florijn et al., 2014), wrinkling (Terwagne et al., 2014; Danas and Triantafyllidis, 2014), fingering (Biggins et al., 2013) or buckling (Mullin et al., 2007; Shim et al., 2012; Coulais et al., 2015) have been created. Collectively they constitute a promising route to develop mechanical devices for sensing (Brenner et al., 2003; Coulais et al., 2016), actuation (Wang et al., 2012; Li et al., 2013; Terwagne et al., 2014; Overvelde et al., 2015) or soft robotics (Autumn et al., 2000; Shepherd et al., 2011).

13 These structures harness postinstabilities *and* their constituents undergo
14 large deformations. A theoretical description of this regime, where as we will
15 show nonlinearities are key, is not well developed yet. On the one hand, the de-
16 scription of postbuckling behaviour has been widely investigated, but for models
17 in which the constitutive material is assumed to be linearly elastic under small
18 deformations (Hutchinson and Koiter, 1970; Budiansky, 1974; Davies et al.,
19 1994; Magnusson et al., 2001; Vaz and Silva, 2003; Mazzilli, 2009; Bažant and
20 Cedolin, 2010; Humer, 2013). On the other hand, much attention has been
21 devoted to characterizing the instabilities of nonlinear elastic cellular materi-
22 als (Geymonat et al., 1993; Lopez-Pamies and Castañeda, 2006b,a; Michel et al.,
23 2007) or structures (Goriely et al., 2008), but only for the onset of instability,
24 and not for the postinstability regime.

25 Euler buckling, known as the phenomenon where an elastic beam will buckle
26 under a sufficiently large compressive axial load, is perhaps the simplest and the
27 most widespread instability (Euler, 1774). Much theoretical attention has been
28 devoted to describing it using the classical (Timoshenko and Gere, 1961; Reiss,
29 1969), extensible and shearable (Antman, 1972) elastica problem. Further in-
30 depth studies have focused on the onset of buckling, the structure of buckled
31 states (Antman and Rosenfeld, 1978; Antman and Pierce, 1990), closed form
32 solutions (Goto et al., 1987, 1990; Pflüger, 2013), large deformations (Simo
33 and Vu-Quoc, 1988; Wang, 1997; Irschik and Gerstmayr, 2009) and three-
34 dimensional (Reissner, 1973; Simo, 1985; Simo and Vu-Quoc, 1986, 1991) defor-
35 mations. In this paper we investigate the postbuckling regime of wide beams,
36 where strains are necessarily large. A salient feature of buckling of slender
37 beams is that the postbuckling compliance increases tremendously after buck-
38 ling, yet remains positive. However, in recent work we showed that wide beams
39 that buckle and undergo large deformations can exhibit a negative postbuckling
40 compliance (Coulais et al., 2015). Although negative compliance is commonly
41 observed in buckling of shells (Bažant and Cedolin, 2010), pipes (Hutchinson
42 and Koiter, 1970) and the wrinkling of membranes (Pocivavsek et al., 2008;
43 Diamant and Witten, 2011; Audoly, 2011), it has not been reported for beam
44 buckling, and to the best of our knowledge is not predicted by existing beam
45 models.

46 Here we develop a 1D nonlinear beam model, that without adjustable param-
47 eters, describes the postbuckling slope of wide neo-Hookean beams. In partic-
48 ular, this model allows to analytically capture the onset of subcritical buckling
49 (postbuckling slope < 0) for widths larger than approximately 15%, in good
50 agreement with experiments and FEM simulations. In Section 2 we expand on
51 our previous experimental and numerical findings to show that for neo-Hookean
52 beams, the postbuckling compliance becomes negative when the beam width-
53 to-length ratio t exceeds approximately 12% (Coulais et al., 2015). In Section 3
54 we construct an effective 1D energy density by combining the Mindlin-Reissner
55 kinematics (Reissner, 1972) with a nonlinearity in the stress-strain relation and
56 establish the governing beam equations. We then solve the beam equations to
57 obtain the variation of the postbuckling slope with t and find that, without any
58 adjustable parameters, our model is in excellent agreement with experiments

59 and simulations. Our work thus unambiguously unravels the link between stress-
60 strain nonlinearity and postbuckling behaviour. While we focus on the buckling
61 of wide neo-Hookean beams, we note that we only need to include quadratic
62 corrections to the stress-strain relation to correctly capture the physics. Hence,
63 for materials with other nonlinear constitutive laws, including metamaterials as
64 explored in Coulais et al. (2015) and Coulais (2016), our description is also valid.
65 Our analytical description can be used to rationally design the postbuckling be-
66 haviour of beams, and we hope that it can inspire work to capture and describe
67 postinstability behaviours of other elastic systems. More widely, our work may
68 impact the design of compliant devices, which harness instabilities (e.g. buck-
69 ling, snapping, wrinkling) to convey mechanical functionalities that are of use
70 in soft robotics (Autumn et al., 2000; Shepherd et al., 2011), sensors (Brenner
71 et al., 2003; Coulais et al., 2016) and actuators (Wang et al., 2012; Li et al.,
72 2013; Terwagne et al., 2014; Overvelde et al., 2015).

73 2. Phenomenology: Subcritical Buckling

74 In this section, we present and expand the findings from our previous work
75 on subcritical buckling of wide beams (Coulais et al., 2015). First, we discuss
76 both the experimental and numerical protocols to study buckling of rectangular
77 beams to determine the force-displacement relation. We consider both the nu-
78 merical protocol for 3D FEM simulations with boundary conditions that closely
79 model the experimental conditions, and 2D simulations with simplified bound-
80 ary conditions. Second, we analyze the onset of buckling and the postbuckling
81 compliance of beams of varying width-to-length ratio t . We then show that for
82 both experiments and numerics the postbuckling compliance varies systemati-
83 cally with t , and becomes negative for $t \gtrsim 0.12$.

84 2.1. Experiments and FEM simulations

85 In the analysis below, we consider beams of the width-to-length ratio $t = w/\ell$
86 and depth d , under load F and corresponding uniaxial displacement u , where
87 $u, F > 0$ correspond to a compressive deformation (Fig. 1a and b).

88 2.1.1. Experiments

89 To perform buckling experiments, we mold 12 solid rectangular beams of rest
90 length $\ell = 45$ mm, depth $d = 35$ mm and widths ranging from $w = 1.55$ mm to
91 $w = 12.85$ mm (Fig. 1(a)) out of a well-characterized silicon rubber (Zhermarck,
92 Polyvinyl Siloxane double elite 8, density 1.15×10^3 kg/m³, Young's modulus
93 $E = 250$ kPa, Poisson's ratio $\nu \approx 0.5$). The extremities of the beams are glued
94 on plexiglass plates that are attached to the uniaxial testing device (Instron
95 5965) in order to approximate clamped-clamped boundary conditions, and we
96 perform the experiments in a water bath in order to limit the effects of gravity.

97 **2.1.2. 3D simulations**

98 We simultaneously carry out a nonlinear analysis using the commercial finite el-
99 element package ABAQUS/STANDARD on beams with the exact same geometry as
100 in the experiments. We determine the buckling point using a specific algorithm
101 in our finite element code that does not require seeding the initial geometry
102 with imperfections (Coulais et al., 2015), allowing to obtain a 0.1% accuracy on
103 the buckling onset.

Material model — The rubbers used in our experiments are well described
by the incompressible neo-Hookean formulation of nonlinear elasticity (Boyce
and Arruda, 2000). We therefore use a neo-Hookean strain energy density (Og-
den, 1997) of the form

$$W = \frac{G}{2} \left(\det(\mathbf{F})^{-2/3} \text{tr}(\mathbf{F}\mathbf{F}^T) - 3 \right) + \frac{K}{2} (\det(\mathbf{F}) - 1)^2, \quad (1)$$

104 where G is the shear modulus, K the bulk modulus and $\mathbf{F} \equiv \partial\mathbf{x}/\partial\mathbf{X}$ is the
105 deformation gradient tensor from the undeformed coordinates \mathbf{X} to the deformed
106 coordinates \mathbf{x} . In the numerical analysis, we use the moduli $G = 83$ kPa and
107 $K = 42$ GPa, which models accurately the $E = 250$ kPa nearly-incompressible
108 rubber used in the experiments.

109 **Boundary conditions** — We numerically impose clamped-clamped bound-
110 ary conditions to resemble the experiments where the endpoints of the beam
111 are glued on plexiglass plates.

112 **2.1.3. Simplified 2D FEM simulations**

113 In addition, we carry out 2D plane stress simulations (Abaqus element type
114 CPS4) using the same material model, yet with simplified slip boundary con-
115 ditions at both endpoints of the beam, which allows for free lateral expansion
116 at the clamped-clamped endpoints to avoid barreling effects (Narayanasamy
117 et al., 1988). The choice for plane stress over plane strain conditions is a priori
118 not obvious because our beams are intermediate between the plane stress limit
119 ($w \gg d$), and plane strain limit ($w \ll d$). We therefore used our 3D simula-
120 tions to investigate the 3D stresses and strains for beam thicknesses where the
121 postbuckling slope changes sign ($t \approx 0.1$). We found that in this case there
122 are significant out of plane strains, but that the out of plane stresses are small
123 (ratio between the lateral and uniaxial stresses < 0.1) – this motivates us to
124 focus on the plane stress case. The plane stress condition, which is nontrivial in
125 finite-strain elasticity, is implemented by requiring that the yy-component of the
126 true (Cauchy) stress is zero, which necessitates the iterative computation of the
127 deformation gradient component F_{yy} to satisfy this condition (Doghri, 2013).
128 Altogether, these assumptions ensure that more complex 3D and boundary ef-
129 fects can be neglected and allow us to carry out the analysis in the simplest
130 setting where subcritical buckling can be observed, and will be used later to

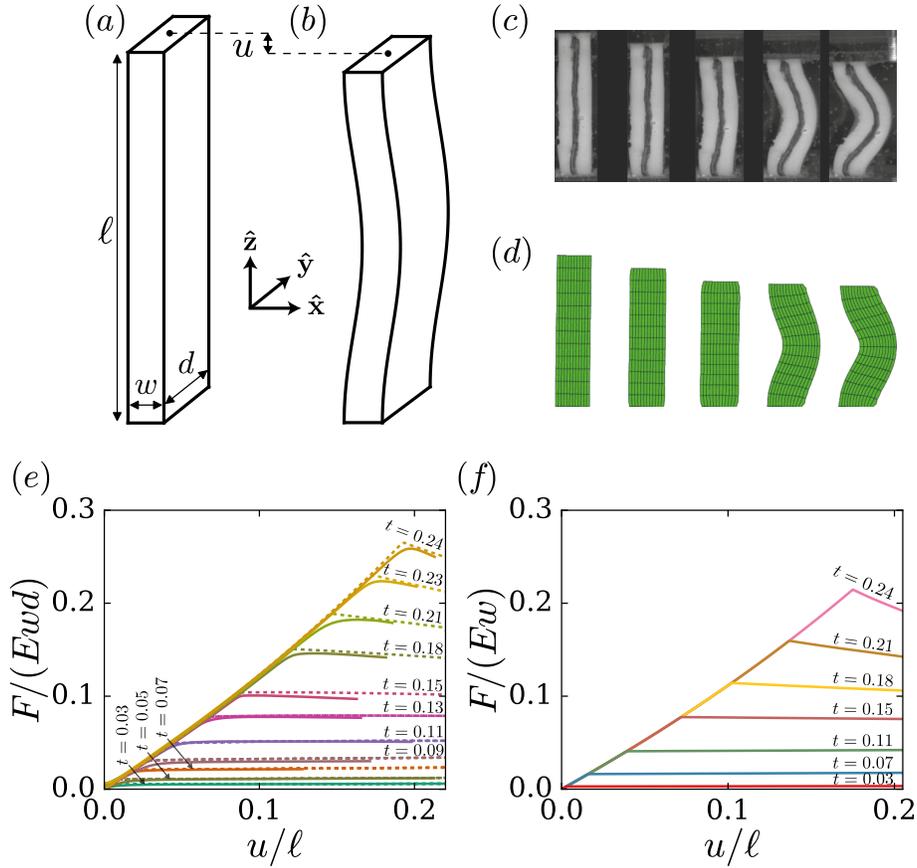


Figure 1: Buckling of wide neo-Hookean beams. (a) Sketch of a beam in its initial undeformed state, for which the beam has a rest length ℓ , width w and depth d . (b) Applying a compressive displacement u , leads to compression and eventually buckling of the beam. (c-d). Frontview snapshots of (c) the experiment and (d) the simulation for a beam of length $\ell = 45$ mm, depth $d = 35$ mm and width $w = 11.95$ mm, at compressive displacements (from left to right) $u = 0$, $u = 0.5 u_c$, $u = 0.99 u_c$, $u = 1.1 u_c$ and $u = 1.2 u_c$. (e-f) Scaled compressive force $F/(Ewd)$ vs. compressive displacement u/ℓ for beams of different width for (e) the experiments (solid lines) and 3D simulations (dashed lines) and (f) the simplified 2D simulations. As the effects of gravity are negligible in the experiments and absent in simulations, the choice of the Young's modulus E is immaterial and we trivially scale the forces by E .

131 pinpoint the physical mechanism at stake in the postbuckling behaviour of wide
 132 beams.

133 2.2. Buckling and Subcritical Buckling

134 In Fig. 1(c-d) we simultaneously display 5 frontview snapshots of experiments
 135 and 3D simulations for a beam with $t = 0.23$ ($w = 10.20$ mm) at different com-
 136 pressive displacements, which are in very good qualitative agreement. Moreover,

137 we plot the obtained force-displacement curves for the complete range of beam
 138 widths in Fig. 1(e), which illustrates that 3D simulations and experiments are
 139 also in very good quantitative agreement. Hence, the neo-Hookean material
 140 model describes the buckling of wide beams well. For all curves, we observe
 141 a near-linear increase until the onset of buckling, at which the slope abruptly
 142 changes. For thin beams, the force increases after buckling, while for thick
 143 beams, the force decreases. For buckling experiments under controlled force of
 144 a sufficiently wide beam, the postbuckling branch would thus be unstable and
 145 the pitchfork instability would be subcritical. Therefore, we refer to this type of
 146 instability as *Subcritical Buckling*. The 2D simulations, albeit considerably simpler,
 147 display qualitatively similar behaviour (Fig. 1(f)), which demonstrates that
 148 subcritical buckling does originate neither from boundary-induced singularities
 149 nor from 3D effects. To the best of our knowledge, although subcritical buckling
 150 is fairly common in other settings such as the wrinkling instability (Moon et al.,
 151 2007; Huang et al., 2007; Cao and Hutchinson, 2011) and the wrinkle-to-fold
 152 transition (Pocivavsek et al., 2008; Diamant and Witten, 2011; Audoly, 2011),
 153 such sign change is not predicted by any theory as of now for the Euler buckling
 154 of wide beams for realistic aspect ratios. Note that Magnusson et al. (2001)
 155 predicted such transition from supercritical to subcritical postbuckling, yet for
 156 overly large aspect ratios ($t=0.24$), and for which the validity of the extensible,
 157 non-shearable elastica is not guaranteed.

We now retrieve the onset of buckling u_c and the postbuckling slope S ,
 using the relation between the load F and the compressive displacement u
 in the postbuckling regime:

$$\frac{F - F_c}{F_c} = S \frac{(u - u_c)}{\ell} + \mathcal{O}\left((u - u_c)^2\right), \quad (2)$$

158 with F_c the critical buckling force. In Fig. 2(a) we display the onset of buckling
 159 as a function of the beam width-to-length ratio t , for the experiments, 3D FEM
 160 simulations and the 2D FEM simulations, and observe quantitative agreement
 161 with the prediction of Euler's elastica for clamped-clamped boundary condi-
 162 tions, $u_c^{euler}/\ell = t^2\pi^2/3$ (Bažant and Cedolin, 2010). While the onset shows
 163 quantitative agreement with Euler's prediction, the results in Fig. 2(b) show
 164 the postbuckling slope S strongly deviates from Euler's prediction $S = 1/2$ as t
 165 increases, and becomes *negative* for $t \gtrsim 0.12$. Importantly, Fig. 2(b) illustrates
 166 that subcritical buckling of wide beams is a robust phenomena: Even with the
 167 simplifications made in the 2D simulations, the differences in the postbuckling
 168 slope between 2D and 3D simulations are modest.

169 The emergence of subcritical Euler buckling is, as we will show, readily
 170 related to nonlinearity in the stress-strain relation (Coulais et al., 2015). In the
 171 following, we will rationalize such behaviour by constructing a 1D model that
 172 encompasses such a stress-strain nonlinearity.

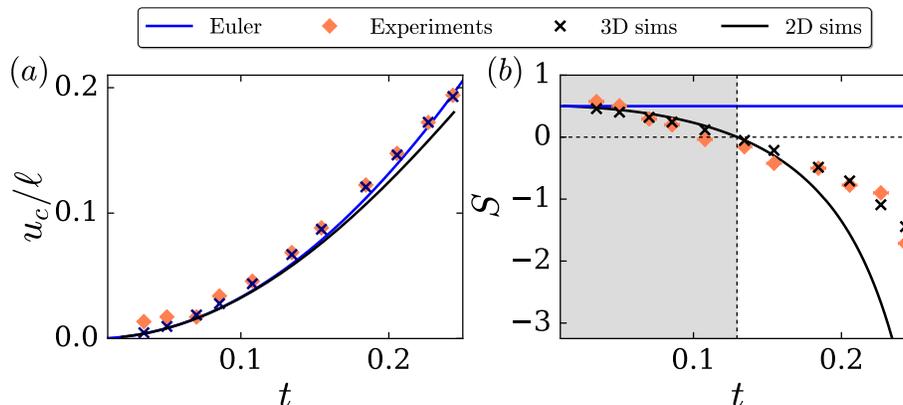


Figure 2: Critical compressive displacement and postbuckling slope as function of the beam width-to-length ratio, for Euler’s elastica (solid blue), experiments (orange diamonds), 3D FEM simulations (black crosses) and 2D plane stress FEM simulations (solid black). (a) The onset of buckling, u_c , in experiments and simulations qualitatively follows Euler’s elastica. (b). The postbuckling slope S in experiments and simulations progressively deviates from the Euler limit $S = 1/2$ for large t . Subcritical buckling ($S < 0$) occurs for $t \gtrsim 0.12$, indicated by the shaded region.

173 3. 1D nonlinear beam model

174 In this section we formulate a 1D nonlinear model to describe the postbuckling
 175 of wide beams. Our model assumes (i) that the kinematics of the 1D model are
 176 captured by the Mindlin-Reissner strains, namely axial strain, curvature and
 177 shear (Reissner, 1972); (ii) that axial stress and strain are related nonlinearly.
 178 Based on these assumptions, we derive an expression for the 1D energy density
 179 as well as the governing equations for the mechanical equilibrium of wide beams.
 180 We then analytically solve the governing equations and find excellent agreement
 181 with 2D simulations for the postbuckling behaviour, without any adjustable
 182 parameters. Finally, we refine our beam model using extensive 2D simulations
 183 and show that distortions from Mindlin-Reissner kinematics have a negligible
 184 effect on the predictions by the model.

185 3.1. Mindlin-Reissner beam with a nonlinear stress-strain 186 relation

Mindlin-Reissner kinematics describe beams that can be compressed, bent and sheared. These three deformation modes are quantified by a compressive $\tilde{\varepsilon}_0(s)$, curvature $\varepsilon_1(s) \equiv \theta_s(s)$ and shear strain $\gamma_0(s)$, as function of the curvilinear coordinate s along the beam’s central axis, with θ the deflection angle of the beam’s axis with respect to the vertical. Therefore the total elastic energy of

these beams is a functional of the form

$$\mathcal{E} [\tilde{\varepsilon}_0(s), \theta(s), \varepsilon_1(s), \gamma_0(s)] = \int_0^\ell ds \epsilon [s, \tilde{\varepsilon}_0(s), \theta(s), \varepsilon_1(s), \gamma_0(s)], \quad (3)$$

187 where the 1D energy density of the beam $\epsilon [s, \tilde{\varepsilon}_0(s), \theta(s), \varepsilon_1(s), \gamma_0(s)]$ exclusively
188 depends on these strains.

The second key assumption is that stress and strain are related nonlinearly. To describe the vicinity of postbuckling, we set up an expansion of the nominal stress σ around the buckling strain ε_b up to quadratic order, which yields:

$$\frac{\sigma - \sigma_b}{E_b} = (\varepsilon - \varepsilon_b) + \eta (\varepsilon - \varepsilon_b)^2 + \mathcal{O}(\varepsilon - \varepsilon_b)^3, \quad (4)$$

189 where E_b and σ_b are the effective Young's modulus and nominal stress at buckling. In the case of neo-Hookean materials under plane stress conditions, the
190 coefficients of this expansion can be determined analytically and read $\eta =$
191 $-1 + \mathcal{O}(\varepsilon_b)$ and $E_b = E + \mathcal{O}(\varepsilon_b)$ (See Appendix A.3 for a demonstration).
192 In the case of plane strain conditions, not considered here, it can be shown that
193 $\eta = -3/2 + \mathcal{O}(\varepsilon_b)$ (See SI in Coulais et al. (2015)). The nonlinearity of the
194 above stress-strain relation stems from the combination of large deformations
195 and incompressibility and can qualitatively be understood from the fact that
196 upon compression (tension) the cross-sectional area increases (decreases) and
197 the stress-strain curve is therefore effectively stiffening (softening). In addition,
198 we assume a linear relation between the nominal shear stress τ and shear strain
199 γ , $\tau = G\gamma$ in agreement with the elasticity of neo-Hookean materials (Ogden,
200 1997).
201

Based on these two assumptions, we find that the 1D energy density describing postbuckling reads:

$$\begin{aligned} \epsilon [\varepsilon_0(s), \varepsilon_1(s), \gamma_0(s)] = & E_b A \varepsilon_b \varepsilon_0 + \frac{1}{2} E_b A \varepsilon_0^2 \\ & + E_b I \left(\frac{1}{2} + \eta \varepsilon_0 \right) \varepsilon_1^2 \\ & + \frac{GA}{2} \gamma_0^2, \end{aligned} \quad (5)$$

202 with $\varepsilon_0(s) = \tilde{\varepsilon}_0(s) - \varepsilon_b$, $A = wd$ (the cross-sectional area) and G is the shear
203 modulus. Crucially, the nonlinear correction proportional to η introduces a
204 nonlinear coupling between the compression strain and the curvature $\varepsilon_0 \varepsilon_1^2$, and
205 such coupling is absent in previous linear beam models (Magnusson et al., 2001;
206 Humer, 2013).

To establish the governing beam equations, the total elastic energy \mathcal{E} has to be minimized under the geometrical constraint set by the boundary conditions. In the case of Euler buckling, a uniaxial displacement is applied along the vertical axis of the beam and is associated to the following geometrical constraint:

$$\Pi = F \left(u - \left(\ell - \int_0^\ell ds ((1 + \varepsilon_b + \varepsilon_0) \cos \theta - \gamma_0 \sin \theta) \right) \right), \quad (6)$$

where F is the Lagrange parameter associated with the axial displacement u that corresponds to the external axial force applied on the beam. We use the fact that $\varepsilon_1 \equiv \theta_s$ to apply the Euler-Lagrange formulation (Marion, 2013) on the energy functional including the constraint:

$$\tilde{\mathcal{E}} [\varepsilon_0(s), \theta(s), \varepsilon_1(s), \gamma_0(s)] = \int_0^\ell ds \epsilon - \Pi, \quad (7)$$

which yields the governing equations of the beam:

$$E_b I \theta_{ss} + F \{(1 + \varepsilon_b + \varepsilon_0) \sin \theta + \gamma_0 \cos \theta\} + 2\eta E_b I (\theta_s \varepsilon_0)_s = 0, \quad (8a)$$

$$F \cos \theta + E_b A (\varepsilon_b + \varepsilon_0) + \eta E_b I \theta_s^2 = 0, \quad (8b)$$

$$GA \gamma_0 - F \sin \theta = 0. \quad (8c)$$

207 This set of three coupled equations determine the beam's central axis in the
 208 postbuckling regime of wide beams. We will refer to this set of equations as the
 209 1D *nonlinear* beam model, since it includes the nonlinearity η .

210 Please note that in the limit of linear materials ($\eta = 0, E_b = E$), Eqs. (8)
 211 correspond to the equations for a shearable and extensible beam derived by
 212 (Humer, 2013). If additionally the beam is assumed non-shearable, $\gamma_0(s) = 0$
 213 and Eq. (8c) drops out, leaving us with a simpler model derived by (Magnusson
 214 et al., 2001). Finally, for inextensible beams $\varepsilon_0(s) = \varepsilon_b = 0$, Eq. (8b) drops out,
 215 and we recover Euler's elastica $EI \theta_{ss} + F \sin \theta = 0$ (Euler, 1774). Our beam
 216 model thus correctly captures all these linear models.

217 3.2. Solutions to the 1D nonlinear beam model

218 In this section we solve the 1D nonlinear beam model given in Eqs. (8) and
 219 show that the postbuckling slope is dramatically changed and the compressive
 220 Mindlin-Reissner strain significantly improved, when incorporating a nonlinear-
 221 ity η .

222 3.2.1. Dimensionless form

The results below will be presented in dimensionless form and we introduce the following dimensionless quantities:

$$\bar{s} = \frac{s}{\ell}; \quad \bar{F} = \frac{F \ell^2}{E_b I}; \quad \Lambda^{-2} = \frac{I}{A \ell^2}. \quad (9)$$

The quantities \bar{s} and \bar{F} represent the dimensionless curvilinear coordinate and force respectively, and $\Lambda \sim \ell/w$ can be recognized as the slenderness ratio (Bažant and Cedolin, 2010). Using the dimensionless quantities, the set of scaled beam equations given in Eqs. (8) reads:

$$\theta_{\bar{s}\bar{s}} + \bar{F} \{(1 + \varepsilon_b + \varepsilon_0) \sin \theta + \gamma_0 \cos \theta\} + 2\eta (\theta_{\bar{s}} \varepsilon_0)_{\bar{s}} = 0, \quad (10a)$$

$$\varepsilon_0 = - (\bar{F} \Lambda^{-2} \cos \theta + \eta \Lambda^{-2} \theta_{\bar{s}}^2 - \varepsilon_b), \quad (10b)$$

$$\gamma_0 = \bar{F} \Lambda^{-2} \frac{E_b}{G} \sin(\theta). \quad (10c)$$

In the remainder of the paper we drop the overbars, unless if noted otherwise. For convenience, we additionally define:

$$r \equiv \frac{E_b}{G} = 2(1 + \nu) + \mathcal{O}(\varepsilon_b(t)), \quad (11)$$

223 where ν is the Poisson ratio.

224 In Eqs. (10) we use Euler's prediction for ε_b , that accurately describes the
 225 onset of buckling, even for wide beams (see Fig. 2(a)). Furthermore, all the
 226 parameters E_b , r and η can be determined theoretically to leading order in
 227 the beam width-to-length ratio t . In what follows we use these predictions as
 228 input parameters and solve Eqs. (10) to obtain a closed-form expression for the
 229 postbuckling slope as function of and to leading order in t .

230 3.2.2. Closed-form expression for the postbuckling slope as a function 231 of t .

Here we derive our main result, namely the postbuckling slope as a function of beam width-to-length ratio t . In deriving the postbuckling slope, we are interested only in the mechanical response of the beam infinitesimally beyond buckling. Therefore, we only need to solve Eq. 10 for small ($\theta(s) \ll 1$), yet nonlinear beam deflections. As a first step, we expand the governing beam equations up to the cubic order in θ , and substitute Eqs. (8b-8c) into Eq. (8a) to obtain:

$$\begin{aligned} 0 = & \theta_{ss} \left(1 - 2\eta (F\Lambda^{-2} + \varepsilon_b) - 6\theta_s^2 \Lambda^{-2} \eta^2 \right) + \theta \left(F + (r-1) F^2 \Lambda^{-2} \right) \\ & - \theta^3 \left(\frac{1}{6} F + \frac{2}{3} (r-1) F^2 \Lambda^{-2} \right) + \theta^5 \left(\frac{1}{12} F^2 \Lambda^{-2} (r-1) \right) \\ & + \left(\theta^2 \theta_{ss} + \theta \theta_s^2 + \frac{1}{6} \theta^3 \theta_s^2 \right) F \Lambda^{-2} \eta. \end{aligned} \quad (12)$$

We now solve this linearized equation using a perturbative expansion that is consistent with the symmetry of Eq. (12), which only contains odd powers in θ , and that matches the imposed clamped-clamped boundary conditions, $\theta(0) = \theta(1) = 0$:

$$\theta(s) = \alpha \sin 2\pi s + \beta \sin 6\pi s. \quad (13)$$

232 Here, α and β physically correspond to the maximum deflection angle of the first
 233 and third harmonic of the Fourier series which describe the beam shape $\theta(s)$.
 234 To see how α and β are coupled, we substitute the perturbative expansion for
 235 $\theta(s)$ in Eq. (12). By collecting all terms proportional to $\sin(6\pi s)$, and setting
 236 the sum of their coefficients to zero, we found that β is coupled to a higher
 237 power of α , specifically $\beta \sim \alpha^3$. Therefore, since $\alpha \ll 1$, $\beta \ll \alpha$, and in the
 238 following we set $\beta = 0$.

Under the assumption $\beta = 0$, Eq. (12) leads to an explicit equation relating the force F to the deflection α . Expanding $F(\alpha)$ for small deflection α , yields the shape of the pitchfork bifurcation (Guckenheimer and Holmes, 1983):

$$F(\alpha, \Lambda, \eta, r) = F_c + \kappa \alpha^2 + \mathcal{O}(\alpha^4), \quad (14)$$

where κ is the curvature of the pitchfork. To connect this excess force to the axial displacement u , we establish the relation between the deflection angle α and the axial displacement using the geometrical relation

$$u/\ell = 1 - \int_0^1 ds \{(1 + \varepsilon_b + \varepsilon_0) \cos \theta - \gamma_0 \sin \theta\}, \quad (15)$$

which upon small deflections, can be expanded to obtain the desired relation $u(\alpha, F, \Lambda, \eta, r)$. We then invert this relation to $\alpha(u, F, \Lambda, \eta, r)$ and substitute it in Eq. (14), resulting in an equation that needs to be solved for $F(u, \Lambda, \eta, r)$. The final step is then to expand the solution for F in the limit $u \rightarrow u_c^+$, which leads to an equation of the form as in Eq. (2), with the postbuckling slope S equal to:

$$S = \frac{1}{2} - \left(\frac{1}{12} + 2\eta^2 \right) \pi^2 t^2 + \mathcal{O}(t^4). \quad (16)$$

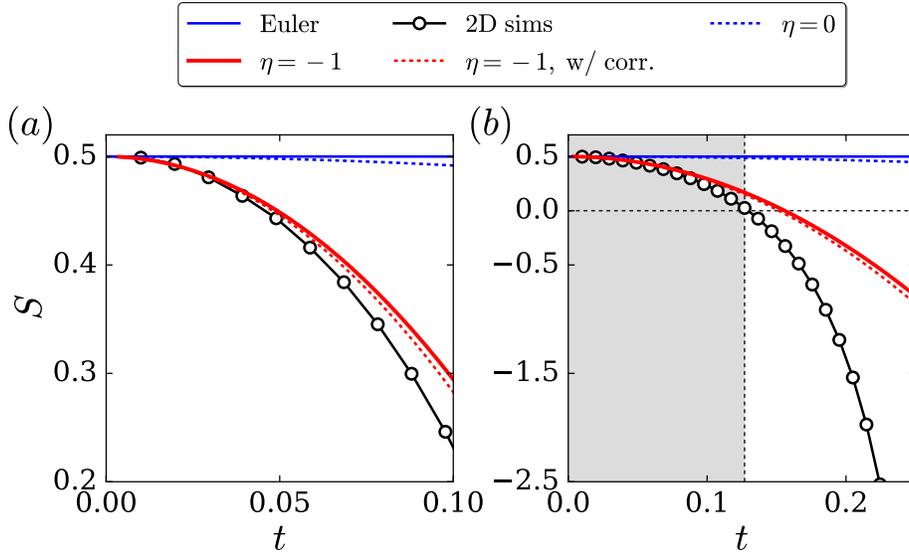


Figure 3: Postbuckling slope S as function of the beam width-to-length ratio t , for five different models. In the Euler limit $S = 1/2$, while in 2D simulations (open circles) S varies with t . Solutions to our model given in Eq. (16) are shown for $\eta = 0$ (dashed blue) and $\eta = -1$ (solid red). Finally, we also show data for an extension of our model discussed in Section 3.3 (dashed red). (a-b) Panel (a) shows a closeup for $0 < t < 0.10$ and panel (b) shows a wider range of width-to-length ratio ($0 < t < 0.25$). The shaded region indicates the cross-over to subcritical buckling ($S < 0$) for the 2D simulations.

239 This result confirms that Euler’s elastica prediction ($S = 1/2$) is recovered in the
240 limit of slender beams ($t \rightarrow 0$) and shows that the leading order correction to the
241 postbuckling slope S is quadratic in t . Notice that such correction comprises the
242 stress-strain nonlinearity η . Does this correction bring an improvement for the
243 prediction of the postbuckling slope? To check this, we compare the value of the
244 postbuckling slope S obtained from 2D simulations to the prediction of Eq. (16),
245 where the value of η is independently determined using the neo-Hookean model
246 under the simplifying assumption that the neo-Hookean material is uniaxially
247 compressed (see Appendix A.3). The comparison shown in Fig. 3 shows excellent
248 agreement between the simulations and our prediction in Eq. (16), namely the
249 quadratic correction matches the data very well for small t and remains accurate
250 up to $t \approx 0.1$ (see Fig. 3(a)). Although we should not expect our prediction
251 to be accurate for wider beams, it remains in qualitative agreement with the
252 simulations and succeeds in predicting subcritical buckling at a critical width-
253 to-length ratio $t \approx 0.15$ (see Fig. 3(b)).

254 Beyond the success of our asymptotic approach, a closer inspection of the
255 quadratic correction to the postbuckling slope S in Eq. (16) allows us to infer
256 three important conclusions. First, the quadratic correction is independent of
257 the ratio of moduli r , given in Eq. (11). Since r sets the magnitude of shear
258 deformations with respect to uniaxial compression, we conclude that shear is
259 subdominant in the lowest order terms of $S(t)$. Second, the coefficient of the
260 quadratic correction is quadratic in η (see Fig. 5), suggesting the sign of the
261 nonlinearity does not play a role. This is consistent with earlier simulations
262 and experiments (Coulais et al., 2015) where we designed metabeams for which
263 $\eta > 0$, in contrast to the neo-Hookean stress-strain nonlinearity for which $\eta < 0$,
264 and found that also in this case S decreases with t . Third, the coefficient of
265 the quadratic correction confirms our initial hypothesis that the stress-strain
266 nonlinearity is the crucial ingredient to capture $S(t)$ correctly: the magnitude
267 of this coefficient is entirely determined by the nonlinearity parameter η . In
268 the absence of η the magnitude of the coefficient is much smaller, and $S(t)$
269 would be only weakly decreasing with t (see Fig. 3). We thus conclude that
270 the nonlinearity η ensures that our theoretical prediction in Eq. (16) is able to
271 capture the subcritical buckling at realistic aspect ratios, in contrast to earlier
272 linear theories (Reissner, 1972; Magnusson et al., 2001; Humer, 2013).

273 **3.2.3. Mindlin-Reissner strains in the nonlinear beam model**

274 We will now illustrate that the prediction for the compressive Mindlin-Reissner
275 strain $\varepsilon_0(s)$ is significantly improved by the nonlinearity η . In Fig. 4(a-c) we
276 plot the compressive, bending and shear Mindlin-Reissner strain for the 2D sim-
277 ulations and the beam model in Eq. (8). First, panel (a) shows a significant
278 qualitative difference in the Mindlin-Reissner strain $\varepsilon_0(s)$ between the linear
279 and nonlinear beam model. In contrast to the linear beam model, the nonlinear
280 beam model is in good qualitative agreement with the FEM simulations and the
281 prefactors of the sinusoidal modulations all carry the same sign, albeit with a
282 slightly smaller amplitude. This confirms our earlier assertion that the nonlin-

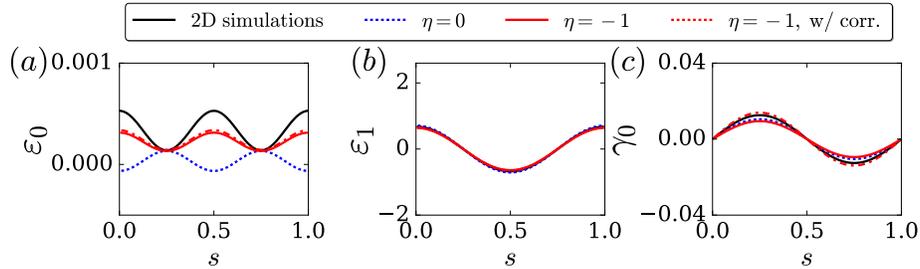


Figure 4: Mindlin-Reissner strains as a function of s , for 4 different models. We consider a wide ($t = 0.1$) beam which is compressed to an axial displacement of $u/u_c = 1.06$. We show results for 2D simulations (solid black), and compare them to numerical solutions to our beam model in Eqs. (8) for $\eta = 0$ (dashed blue) and $\eta = -1$ (solid red). Finally, we also show numerical solutions to an extension of our beam model in Eqs. (20) discussed in Section 3.3 (dashed red). (a-c) We have respectively plotted the compressive, bending and shear Mindlin-Reissner strain along the beam.

283 earity η is the crucial factor to capture correctly the large deformations of wide
 284 neo-Hookean beams. Finally, panel (b) and (c) show that the Mindlin-Reissner
 285 strains $\varepsilon_1(s)$ and $\gamma_0(s)$ remain essentially unchanged due to the nonlinearity and
 286 the model shows excellent agreement with the 2D simulations.

287 3.3. Distortions from Mindlin-Reissner kinematics with non- 288 linear stress-strain relation

289 The previous derivation of the 1D nonlinear beam model in Eqs. (8) is simple
 290 and directly follows from the use of two basic assumptions. In particular, using
 291 Mindlin-Reissner kinematics is a customary yet not controlled assumption. In
 292 this section, we investigate the validity of such a choice by using extensive nu-
 293 merical simulations and demonstrate that distortions from the Mindlin-Reissner
 294 kinematics systematically occur, modifying the 1D energy density and governing
 295 equations, albeit with a subdominant effect.

To explore deviations from Mindlin-Reissner strains, we investigate system-
 atically the stress and strain profiles in Appendix A. In particular, we find that
 the axial strain profile at the center of the beam takes the form:

$$\varepsilon(x) - \varepsilon_b = \tilde{\varepsilon}_0 + \varepsilon_1 x + \varepsilon_2 x^2 + \varepsilon_3 x^3 + \dots, \quad (17)$$

296 where $x \in [-\frac{w}{2}, \frac{w}{2}]$ is the transverse coordinate across the beam width. $\tilde{\varepsilon}_0$ and
 297 ε_1 are Mindlin-Reissner strains introduced in Section 3.1 and ε_i (with $i \geq 2$)
 298 correspond to distortions from a linear axial strain profile. In Appendix A we
 299 have also performed a similar systematic analysis for the shear profile.

Following the extensive simulations and thorough asymptotic analysis pro-

cedure in Appendix A, we find that the 1D energy density takes the form:

$$\begin{aligned} \epsilon [\varepsilon_0(s), \varepsilon_1(s), \gamma_0(s)] = & E_b A \varepsilon_b \varepsilon_0 + \frac{1}{2} E_b A (1 + \zeta_2(\eta)) \varepsilon_0^2 \\ & + E_b I \left(\frac{1}{2} (1 + \zeta_1(t)) + \eta \varepsilon_0 \right) \varepsilon_1^2 \\ & + \frac{GA}{2} \gamma_0^2 (k_1 + k_2 \gamma_0^2), \end{aligned} \quad (18)$$

where the coefficients E_b , η , $\zeta_1(t)$, $\zeta_2(\eta)$, G , k_1 and k_2 can be determined numerically. Note that in the limit when ζ_1 , ζ_2 and k_2 are zero, we recover Eq. (5). Eq. (18) is very similar to Eq. (5) and the numerical values of the coefficients E_b , η , and G match the values that come from the neo-Hookean material model (Ogden, 1997) (see Appendix A). In addition we see that the differences associated to distortions from the Mindlin-Reissner kinematics can be captured by the coefficients $\zeta_1(t)$, $\zeta_2(\eta)$, k_1 and k_2 . While $k_1 = 0.67 \pm 0.15$ is a classical coefficient known as the shear correction factor (Timoshenko, 1921; Timoshenko and Goodier, 1970) whose value quantitatively matches Timoshenko's prediction (Cowper, 1966; Hutchinson, 2000), $\zeta_1(t)$, $\zeta_2(\eta)$, and k_2 are undocumented and correspond to higher order distortions of the strain profiles. They have been determined in Appendix A as:

$$\zeta_1(t) = 6(t^2 + t^4), \quad (19a)$$

$$\zeta_2(\eta) = -0.2 - 0.15\eta, \quad (19b)$$

$$k_2(t) = 0.0013t^{-4}. \quad (19c)$$

300 Note that even though $k_2(t)$ is singular for $t \rightarrow 0$, γ_0 scales as t^4 , such that the
 301 product $k_2 \gamma_0^4$ that arises in Eq. (18) is regularized for $t \rightarrow 0$. Nonetheless, we see
 302 that the distortions in Eqs. (19) introduce minor modifications of the prefactors
 303 in Eq. (18) and in what follows we show that they do not play a major role in
 304 the model.

We now carry out the same Euler-Lagrange approach as previously and find the refined governing equations:

$$\zeta_1(t) E_b I \theta_{ss} + F \{ (1 + \varepsilon_b + \varepsilon_0) \sin \theta + \gamma_0 \cos \theta \} + 2\eta E_b I (\theta_s \varepsilon_0)_s = 0, \quad (20a)$$

$$F \cos \theta + E_b A (\varepsilon_b + \zeta_2(\eta) \varepsilon_0) + \eta E_b I \theta_s^2 = 0, \quad (20b)$$

$$GA \gamma_0 (k_1 + 2k_2 \gamma_0^2) - F \sin \theta = 0. \quad (20c)$$

This set of equations is the equivalent of the previously established Eqs. (8a-8c) and has been determined through a well defined and rigorous set of assumptions. Unfortunately, the coefficients $\zeta_1(t)$, $\zeta_2(\eta)$ and k_2 have to be determined numerically. Following the procedure in Section we linearize and solve Eqs. (20) and find that

$$S = \frac{1}{2} + \frac{(-3 + 2(1 + \zeta_2(\eta)) - 24\eta^2) \pi^2}{12(1 + \zeta_2(\eta))} t^2 + \mathcal{O}(t^4), \quad (21)$$

305 which reduces to Eq. (16) by setting $\zeta_2 = 1$. We have plotted Eq. (21) in Fig. 3
 306 and see that the corrections ζ_1, ζ_2 and k_2 result in a minor improvement to the
 307 postbuckling prediction. Finally, we numerically solved Eqs. 20 to obtain the
 308 Mindlin-Reissner strains and plotted the result for $\eta = -1$ in Fig. 4. Again,
 309 we find that the corrections result in a minor improvement to the postbuckling
 310 prediction. Altogether, this illustrates that the corrections ζ_1, ζ_2 and k_2 have a
 311 subdominant contribution to the postbuckling behaviour.

312 4. Conclusions and discussion

313 We have presented a thorough investigation of the postbuckling of nonlinear
 314 elastic beams, using experiments, finite element simulations and theory. In par-
 315 ticular we have focussed on subcritical buckling, where, for neo-Hookean beams,
 316 the slope of the force-displacement curve becomes negative beyond buckling
 317 when the beam width-to-length ratio exceeds 12%. The main result of this pa-
 318 per is a 1D nonlinear beam model that includes a material nonlinearity η . We
 319 constructed the model by building the beam's energy density using Mindlin-
 320 Reissner kinematics with a nonlinearity in the stress-strain relation, and demon-
 321 strated that this nonlinearity is crucial to accurately capture the postbuckling
 322 behaviour of wide beams and in particular to predict subcritical buckling. In
 323 contrast with previous works that have reported a significant effect of the ra-
 324 tio E/G on the flexure response (Goto et al., 1990) and the critical buckling
 325 force (Humer, 2013) of extensible and shearable beams, we found that E/G has
 326 a subdominant effect on the postbuckling slope.

327 Though our model has been established in the case of neo-Hookean mate-

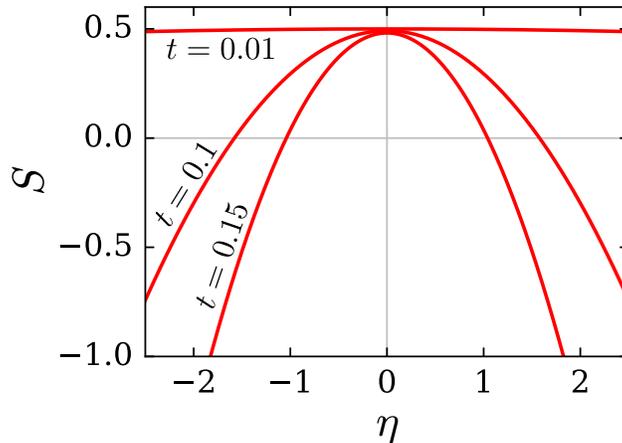


Figure 5: Postbuckling slope as a function of the nonlinearity η . Using Eq. (16) we have plotted $S(\eta)$ for $t = 0.01$, $t = 0.1$ and $t = 0.15$. The curves show that the postbuckling slope is quadratic in η and that the postbuckling slope does not exceed $S = 1/2$.

328 rial nonlinearity ($\eta < 0$), our findings could be generalized to a wider class of
329 nonlinear elastic materials, such as cellular materials with nonlinear effective
330 properties (Gibson and Ashby, 1997; Castañeda, 1991; Coulais, 2016). We ex-
331 pect this generalization to hold provided that the leading nonlinearity of the
332 elastic material is quadratic in nature and that the material strains do not sig-
333 nificantly deviate from the Mindlin-Reissner strain decomposition (as is shown
334 in Section 3.3 for 2D plane stress beams). For example, in recent work by
335 Coulais et al. (2015), beams patterned with a periodic 2D pattern of pores were
336 shown to exhibit positive, geometrically induced nonlinearity ($\eta > 0$). They
337 found that a sufficiently strong nonlinearity leads to subcritical buckling, even
338 when the beam width-to-length ratio is small. Such a transition to subcritical
339 buckling for $\eta > 0$ is in qualitative agreement with our theory that predicts that
340 the postbuckling slope essentially decreases quadratically in η with its maximum
341 at $\eta = 0$ (see Fig. 5). The present work rationalizes those findings and provide
342 strong guidelines for the design of postinstability regimes in soft structures and
343 architected materials (Ashby and Brechet, 2003), where arbitrary values of η
344 can be achieved (Coulais, 2016). We envision in particular that our description
345 could be of interest for the design of compliant hierarchical cellular materials,
346 which often rely on the buckling instability for their functionality (Cho et al.,
347 2014; Yang et al., 2016).

348 In addition, we note that other types of material nonlinearities could be
349 explored and addressed within our framework, for instance, plasticity, stress-
350 relaxation, swelling Yoon et al. (2010); Holmes et al. (2011); Kim et al. (2012);
351 Pezzulla et al. (2015); Na et al. (2016) or even growth and activity, which are
352 ubiquitous in biological solids (Gladman et al., 2016; Sharma et al., 2016)

353 Finally, while our work could be of great use for the engineering of systems
354 that draw on Euler buckling for their functionality (Wang et al., 2014; Shim
355 et al., 2012), a plethora of compliant architected material harness the *snapping*
356 instability (Brenner et al., 2003; Holmes and Crosby, 2007; Shim et al., 2012;
357 Nasto et al., 2013; Florijn et al., 2014; Overvelde et al., 2015; Frenzel et al., 2016;
358 Raney et al., 2016; Coulais et al., 2017). In order to understand the role of ma-
359 terials nonlinearity on such instability and devise mechanical design guidelines,
360 our present framework should be generalized to pre-curved geometries, such as
361 curved beams and shells.

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367 **Appendix A. Asymptotic Analysis of 2D FEM**
 368 **simulations**

369 In this appendix, we use 2D FEM simulations to illustrate and quantify the
 370 role of nonlinearities in the stress-strain relation, set up a systematic series
 371 expansion for the spatial variation of stress and strain across the beam, and use
 372 the numerical results to determine the dominant terms in this expansion. Our
 373 findings will allow us to unambiguously establish a well defined expression for
 374 the 1D energy density of the beam and to compare it with standard limits such
 375 as Euler’s elastic, Timoshenko beams and Mindlin-Reissner beams.

376 **Appendix A.1. Series expansion of the axial nominal stress**
 377 **and strain**

378 First, we perform a systematic expansion of the nominal stress and strain profiles
 379 in the beam’s transverse coordinate x/w , the beam width-to-length ratio t and
 380 the excess displacement $\Delta u \equiv (u - u_c)/u_c$, and determine all prefactors and
 381 scaling exponents using our FEM results.

382

Standard beam theories such as Mindlin-Reissner theory assume that the nominal stress and strain profiles are linear in x . In wide 2D neo-Hookean beams, the deformation field is more complex and we analyze deviations from a linear profile by expanding the nominal strain and stress around the buckling strain (ε_b) and stress (σ_b), as function of the (scaled) transverse coordinate x/w , at the middle of the beam:

$$\varepsilon\left(t, \Delta u, \frac{x}{w}\right) - \varepsilon_b = \sum_{n=0} C_n(t, \Delta u) \left(\frac{x}{w}\right)^n, \quad (\text{A.1a})$$

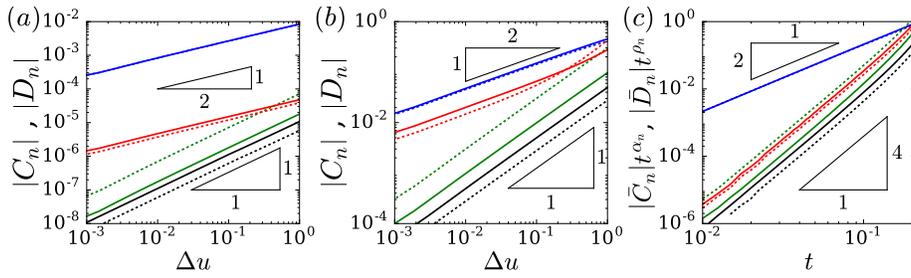


Figure A.6: Expansion of the nominal strain and stress profiles obtained by FEM simulations, according to Eqs. (A.2a-A.2b). We plot the postbuckling profile coefficients C_n and D_n in each order as a function of Δu and t . In black, blue, green and red we have plotted C_n (solid lines) and D_n (dashed lines), corresponding to the order $n = 0$, $n = 1$, $n = 2$ and $n = 3$ respectively. (a-b). We have plotted $|C_n|$ and $|D_n|$ as function of Δu for (a) a slender beam ($t = 0.02$) and (b) a thick beam ($t = 0.15$). (c). Dependence of C_n and D_n on the beam aspect ratio t .

n	Δu		t	
	β_n	τ_n	α_n	ρ_n
0	1.03 ± 0.1	1.05 ± 0.1	4.06 ± 0.4	4.09 ± 0.4
1	0.51 ± 0.03 [$\frac{1}{2}$]	0.51 ± 0.03 [$\frac{1}{2}$]	2.00 ± 0.1 [2]	1.99 ± 0.1 [2]
2	1.03 ± 0.1	1.02 ± 0.1	4.02 ± 0.4	4.00 ± 0.4
3	0.51 ± 0.05	0.51 ± 0.05	4.00 ± 0.4	3.99 ± 0.4
4	1.0 ± 0.15	1.01 ± 0.15	6.05 ± 0.9	6.00 ± 0.9
5	0.51 ± 0.15	0.52 ± 0.16	6.20 ± 1.9	5.82 ± 1.7

Table A.1: Postbuckling profile scaling exponents of Δu and t , for the expansion of the nominal strain and stress profiles as defined in Eqs. (A.1a-A.2b). Each row corresponds to a different order of n and values within the square brackets represent analytical results as predicted by Euler’s elastica for clamped-clamped boundary conditions.

and

$$\frac{\sigma - \sigma_b}{E} \left(t, \Delta u, \frac{x}{w} \right) = \sum_{n=0} D_n(t, \Delta u) \left(\frac{x}{w} \right)^n, \quad (\text{A.1b})$$

where C_n and D_n are the coefficients of the expansion in x/w of order n . We refer to these coefficients as the *postbuckling profile coefficients*. At buckling ($\Delta u = 0$), $C_n = D_n = 0$, so it is natural to assume that the postbuckling profile coefficients C_n and D_n grow as power laws in t and Δu in the postbuckling regime. Therefore, we postulate:

$$C_n(t, \Delta u) = \bar{C}_n t^{\alpha_n} \Delta u^{\beta_n}, \quad (\text{A.2a})$$

and

$$D_n(t, \Delta u) = \bar{D}_n t^{\rho_n} \Delta u^{\tau_n}. \quad (\text{A.2b})$$

383 Here, α_n , β_n , ρ_n and τ_n are *postbuckling profile scaling exponents* and \bar{C}_n and
384 \bar{D}_n are the *postbuckling profile prefactors* which we will now determine up to
385 the order $n = 5$ from our numerical simulations.

386 To determine all the constants, we use the numerical protocol described in
387 Section 2.1.3 and perform $N = 10^2$ simulations for beams with a logarithmically
388 spaced width-to-length ratio in the range from $t = 0.01$ up to $t = 0.25$, and
389 with an excess strain that is increased from $\Delta u = 10^{-3}$ up to $\Delta u = 1$ in
390 3×10^2 subsequent steps. For each simulation we extract the spatial shape
391 of the nominal stress and strain as function of x/w across the middle of the
392 beam at $s = \ell/2$ and fit $\varepsilon(x)$ and $\sigma(x)/E$ to polynomials of order $n = 5$, by
393 which we obtain the postbuckling profile coefficients $C_n(t, \Delta u)$ and $D_n(t, \Delta u)$
394 for each specific set of parameter values t and Δu . From these quantities we
395 subsequently deduce the postbuckling profile scaling exponents and prefactors
396 up to order $n = 5$.

n	\bar{C}_n	\bar{D}_n	\bar{C}_n/\bar{D}_n
0	72.0 ± 25	38.3 ± 13	1.88 ± 0.9
1	$21.3 \pm 3.2 \left[\frac{4\pi^2}{\sqrt{3}} \approx 22.8 \right]$	$21.1 \pm 3.2 \left[\frac{4\pi^2}{\sqrt{3}} \approx 22.8 \right]$	$1.01 \pm 0.2 [1]$
2	-116 ± 41	-553 ± 194	0.21 ± 0.1
3	320 ± 112	254.9 ± 89	1.26 ± 0.6
4	$-6.1 \cdot 10^3 \pm 2.4 \cdot 10^3$	$-1.4 \cdot 10^4 \pm 5.6 \cdot 10^3$	0.42 ± 0.2
5	$1.1 \cdot 10^4 \pm 5.5 \cdot 10^3$	$1.2 \cdot 10^4 \pm 6 \cdot 10^3$	0.99 ± 0.7

Table A.2: Postbuckling profile prefactors \bar{C}_n and \bar{D}_n and their ratio, for the expansion of the nominal strain and stress profiles as defined in Eq. (A.1a-A.2b). Each row corresponds to a different order of n and values within the square brackets represent analytical results, predicted by Euler’s elastica for clamped-clamped boundary conditions.

397 The results of this fitting procedure, shown in Fig. A.6 and Tables (A.1-A.2))
398 confirm the validity of the polynomial asymptotic decomposition Eqs. (A.1a-
399 A.2b). In the following, we carry out a similar analysis for shear deformations.

400

401 Appendix A.2. Series expansion of the nominal shear stress 402 and strain

Second, we investigate shear effects using a similar expansion as above in the beam transverse coordinate x/w , the beam width-to-length ratio t and the excess displacement $\Delta u \equiv (u - u_c)/u_c$, and determine all prefactors and scaling exponents using our FEM results. Standard beam theories such as Mindlin-Reissner theory assume that the nominal shear stress and strain profiles are constant across the beam. In wide 2D neo-Hookean beams, the deformation field is more complex and we analyze deviations from a constant profile by expanding the nominal shear strain and stress around the buckling strain and stress. Following a similar series expansion as in Eqs. (A.1a-A.2b), we expand the nominal shear strain and stress profiles as:

$$\gamma \left(t, \Delta u, \frac{x}{w} \right) = \sum_{n=0} J_n(t, \Delta u) \left(\frac{x}{w} \right)^n, \quad (\text{A.3a})$$

and

$$\frac{\tau}{G} \left(t, \Delta u, \frac{x}{w} \right) = \sum_{n=0} K_n(t, \Delta u) \left(\frac{x}{w} \right)^n, \quad (\text{A.3b})$$

where J_n and K_n are the postbuckling profile coefficients of the expansion at order n . Note that prior to buckling, the beam simply undergoes uniform uniaxial compression and has not developed any curvature yet. Therefore, unlike the uniaxial nominal strain and stress which are constant across the beam in

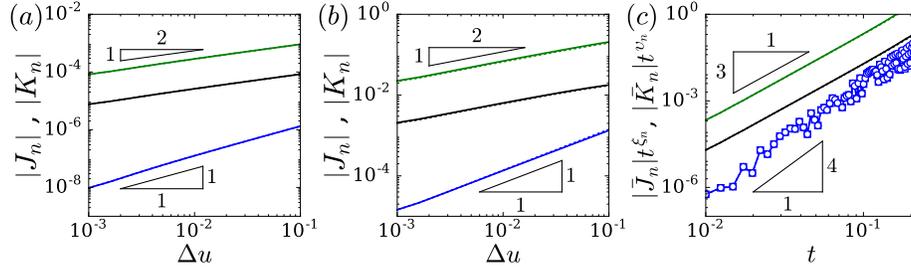


Figure A.7: Dependence of the spatial nominal shear strain and stress profiles on Δu and t , obtained by FEM simulations. In black, blue and green we have plotted the postbuckling profile coefficients J_n (solid lines) and K_n (dashed lines), corresponding to order $n = 0$, $n = 1$ and $n = 2$ respectively. (a-b). We have plotted $|J_n|$ and $|K_n|$ as function of Δu for (a) a slender beam ($t = 0.02$) and (b) a thick beam ($t = 0.15$). (c). Dependence of J_n and K_n on the beam's aspect ratio t .

the prebuckling regime, the shear stress and strain are strictly zero for $\Delta u \leq 0$. Similarly to the postbuckling profile coefficients C_n and D_n (Eqs. (A.2a-A.2b)), we use that $J_n = K_n = 0$ at buckling, and we assume that the postbuckling profile coefficients J_n and K_n grow as power laws in t and Δu in the postbuckling regime:

$$J_n(t, \Delta u) = \bar{J}_n t^{\xi_n} \Delta u^{\Xi_n}, \quad (\text{A.4a})$$

and

$$K_n(t, \Delta u) = \bar{K}_n t^{v_n} \Delta u^{\Upsilon_n}. \quad (\text{A.4b})$$

403

404 Here, ξ_n , Ξ_n , v_n and Υ_n are the postbuckling profile scaling exponents, and
 405 \bar{J}_n and \bar{K}_n are the postbuckling profile prefactors which we determine from
 406 numerical simulations.

n	Δu		t	
	Ξ_n	Υ_n	ξ_n	v_n
0	0.49 ± 0.02	0.49 ± 0.02	3.02 ± 0.15	3.01 ± 0.15
1	1.03 ± 0.05	1.06 ± 0.05	3.98 ± 0.20	3.98 ± 0.20
2	0.50 ± 0.03	0.50 ± 0.03	3.02 ± 0.15	3.01 ± 0.15
3	1.02 ± 0.26	0.93 ± 0.23	5.93 ± 1.48	5.70 ± 1.43

Table A.3: Postbuckling profile scaling exponents of Δu and t , for the expansion of the nominal shear strain and stress profiles as defined in Eq. (A.3a-A.4b). Each row corresponds to a different order of n and results are provided up to cubic order ($n = 3$).

n	\bar{J}_n	\bar{K}_n	\bar{J}_n/\bar{K}_n
0	-18.9 ± 1.9	-18.8 ± 1.9	1.0 ± 0.14
1	-45.3 ± 4.5	-45.3 ± 4.5	1.0 ± 0.14
2	205.9 ± 20.6	204.3 ± 20.4	1.0 ± 0.14
3	$-4.4 \cdot 10^3 \pm 1.8 \cdot 10^3$	$-3.7 \cdot 10^3 \pm 1.5 \cdot 10^3$	1.2 ± 0.68

Table A.4: Postbuckling profile prefactors \bar{J}_n and \bar{K}_n and their ratio, for the expansion of the nominal shear strain and stress profiles as defined in Eqs. (A.3a-A.4b). Each row corresponds to a different order of n and results are provided up to cubic order ($n = 3$).

407 To determine all the constants, we use the same set of $N = 10^2$ FEM sim-
408 ulations as before, from which we now extract the spatial shape of the nominal
409 shear stress and strain as function of x/w along a cross section at one quarter
410 of the beam, $s = \ell/4$, and fit $\gamma(x)$ and $\tau(x)/G$ to polynomials of order $n = 3$.
411 From the resulting fits we then obtain the postbuckling profile coefficients J_n
412 and K_n for a specific set of parameter values t and Δu . From these quantities we
413 subsequently deduce the postbuckling profile scaling exponents and prefactors
414 up to order $n = 3$. The results of this fitting procedure, shown in Fig. A.6 and
415 Tables (A.1-A.2)) confirm the validity of the polynomial asymptotic decompo-
416 sition Eqs. (A.1a-A.2b). In the following, we discuss the implications of such
417 asymptotic analysis for the formulation of 1D models.

418

419 Appendix A.3. Effective stress-strain relations

420 In this appendix we set up the appropriate stress-strain relations for both the
421 uniaxial and shear stress-strain relation.

422 Appendix A.3.1. Nonlinear Uniaxial stress-strain relation

From Table A.2 we see that the coefficients \bar{C}_n and \bar{D}_n are not equal, thus
evidencing a nonlinearity in the stress-strain relation. Because the postbuckling
slope (Eq. 2) is defined in the vicinity of the buckling point, the starting point
is to write a Taylor series for the normal stress around the buckling strain ε_b up
to quadratic order, which yields

$$\frac{\sigma - \sigma_b}{E_b} = (\varepsilon - \varepsilon_b) + \eta (\varepsilon - \varepsilon_b)^2 + \mathcal{O}(\varepsilon - \varepsilon_b)^3. \quad (\text{A.5})$$

We can calculate E_b and η analytically by evaluating the expansion in Eq. (A.5)
using the stress-strain relation for uniaxially compressed neo-Hookean materials
(Ogden, 1997). This yields:

$$E_b(\varepsilon_b) = \frac{E}{3} \left(1 + \frac{2}{(1 + \varepsilon_b)^3} \right), \quad (\text{A.6a})$$

and

$$\eta(\varepsilon_b) = -\frac{3}{2(1 + \varepsilon_b) + (1 + \varepsilon_b)^4}. \quad (\text{A.6b})$$

Eq. (A.6a) and Eq. (A.6b) show that as ε_b becomes increasingly negative, both the effective stiffness E_b and the magnitude of nonlinearity parameter η increase. In particular, we find, by expanding Eqs. (A.6a-A.6b) for small ε_b , that the leading order corrections to E_b and η are linear in ε_b :

$$E_b/E = 1 - 2\varepsilon_b + \mathcal{O}(\varepsilon_b^2), \quad (\text{A.6c})$$

$$\eta = -1 + 2\varepsilon_b + \mathcal{O}(\varepsilon_b^2). \quad (\text{A.6d})$$

Furthermore, note that as $\varepsilon_b \rightarrow 0$, we retrieve $E_b/E = 1$ and $\eta = -1$, in agreement with the small strain limit of uniaxially compressed neo-Hookean materials (Ogden, 1997). Finally, Eq. (A.6d) is also consistent with the value that we can calculate from the numerical constants C_n and D_n using Table A.2, namely:

$$\eta = \frac{D_2 - C_2}{C_1^2} \approx 1 \quad (\text{A.7})$$

423 **Appendix A.3.2. Linear shear stress-strain relation**

In addition, we have seen from Table A.4 that the coefficients \bar{J}_n and \bar{K}_n are equal, therefore the nominal shear strain and stress are linearly related, hence we can assume

$$\tau(x) = G\gamma(x), \quad (\text{A.8})$$

424 which is the result as predicted by Ogden (1997) in the case of simple shear
 425 for neo-Hookean materials. We will use this linear constitutive equation for the
 426 shear in the remainder of this paper.

427 **Appendix B. Construction of the 1D energy den-** 428 **sity comprising stress-strain nonlin-** 429 **earity**

430 In this appendix, we construct the energy density based on Mindlin-Reissner
 431 kinematics and a nonlinear stress-strain relation and take into account distortions
 432 to the Mindlin-Reissner kinematics. This 1D energy density is the base of
 433 our models, presented in Section 3 of the main text.

We start by expressing the total increase of the elastic energy beyond buckling. This increase follows from an integral of the respective products of stress

and strain, integrated over the surface area of the beam, that is,

$$\begin{aligned} \mathcal{E}/d = \int ds dx \left(\int_0^{\varepsilon_{xx}} d\varepsilon'_{xx} \sigma_{xx} + \int_0^{\varepsilon_{yy}} d\varepsilon'_{yy} \sigma_{yy} + \int_{\varepsilon_b}^{\varepsilon_{zz}} d\varepsilon'_{zz} \sigma_{zz} \right. \\ \left. + \int_0^{\varepsilon_{xy}} d\varepsilon'_{xy} \sigma_{xy} + \int_0^{\varepsilon_{yz}} d\varepsilon'_{yz} \sigma_{yz} + \int_0^{\varepsilon_{xz}} d\varepsilon'_{xz} \sigma_{xz} \right). \end{aligned} \quad (\text{B.1})$$

434 Even though we consider 2D beams, we keep a factor d (the depth) here to
 435 facilitate comparison to 3D beam results. For 2D beams, the ‘ yy ’, ‘ xy ’ and
 436 ‘ yz ’ contributions are zero. Moreover, since the beam can freely expand along
 437 the x direction without any barrelling effects near the boundaries, we expect
 438 that $\sigma_{xx} \approx 0$ at each point of the beam, an assumption which we have verified
 439 numerically in our 2D simulations. As a result, we are left with the ‘ zz ’ and ‘ xz ’
 440 terms, which correspond to the uniaxial and shear deformations, respectively.
 441 Our aim is to set up an energy functional using the Mindlin-Reissner strains —
 442 1D fields describing the shape of the beam along the curvilinear coordinate s .

Therefore we define a linear energy density $\epsilon(s)$ as follows:

$$\mathcal{E} = \int_0^\ell ds \epsilon(s), \quad (\text{B.2a})$$

where

$$\frac{\epsilon(s)}{d} = \int_{-\frac{w}{2}}^{\frac{w}{2}} dx \int_{\varepsilon_b}^{\varepsilon(x)} \sigma(\varepsilon') d\varepsilon' + \int_{-\frac{w}{2}}^{\frac{w}{2}} dx \int_{\varepsilon_b}^{\gamma(x)} \tau(\gamma') d\gamma', \quad (\text{B.2b})$$

443 with $\varepsilon \equiv \varepsilon_{zz}$ and $\gamma \equiv \varepsilon_{xz}$. Here, $\epsilon(s)$ represents the linear energy density that
 444 captures the amount of energy in a cross sectional area of the beam per unit
 445 length of the curvilinear coordinate s .

446 Appendix B.1. 1D energy density including distortions from 447 Mindlin-Reissner kinematics

Here we present the energy density comprising distortions from the Mindlin-Reissner strains and built with the aid of the numerical results. To this end, we substitute the respective stress-strain relations (Eq. (A.5) and Eq. (A.8)) to carry out the integration with respect to the nominal strains ε and γ . Second, we integrate with respect to x by using the expansions of the uniaxial and shear strain profiles up to cubic order (Eqs. (A.1a) and (A.3a)). This yields:

$$\begin{aligned} \frac{\epsilon}{E_b A} = & \left\{ \frac{\bar{C}_1^2}{24} \right\} \Delta u t^4 \\ & + \left\{ \frac{\bar{C}_1 \bar{C}_3}{80} + \bar{C}_0 \bar{C}_B + \frac{\bar{C}_2 \bar{C}_B}{12} + \frac{G}{E_b} \left(\frac{1}{2} \bar{J}_0^2 + \frac{1}{12} \bar{J}_0 \bar{J}_2 + \frac{1}{160} \bar{J}_2^2 \right) \right\} \Delta u t^6 \\ & + \left\{ \frac{\bar{C}_3^2}{896} + \left(\frac{\bar{C}_0^2}{2} + \frac{\bar{C}_0 \bar{C}_2}{12} + \frac{\bar{C}_2^2}{160} + \eta \frac{\bar{C}_0 \bar{C}_1^2}{12} + \eta \frac{\bar{C}_1^2 \bar{C}_2}{80} + \frac{G}{24 E_b} \bar{J}_1^2 \right) \Delta u \right\} \Delta u t^8 \\ & + \mathcal{O}(\Delta u^2 t^{10}). \end{aligned}$$

(B.3)

448 We have now established carefully the beam's energy density up to second order
 449 in excess strain and eighth order width-to-length ratio, $\mathcal{O}(\Delta u^2 t^8)$. The above
 450 analysis identifies and quantifies precisely how nonlinearity in stress-strain laws
 451 and distortions to Mindlin-Reissner kinematics alter the 1D energy density for-
 452 mulation. While the order $\mathcal{O}(\Delta u t^4)$ corresponds exactly to Euler's elastica, the
 453 order $\mathcal{O}(\Delta u t^6)$ comprises the classical Timenshenko beam contribution as well
 454 as distortions from the linear bending profile. The order $\mathcal{O}(\Delta u t^8)$ contains the
 455 nonlinearity η as well as further distortions for bending and shear.

After a few manipulations which we explain hereafter, it can be shown that Eq. (B.3) can be converted in terms of the Mindlin-Reissner strains as:

$$\begin{aligned} \frac{\epsilon}{E_b} = A\varepsilon_b \varepsilon_0 + \frac{1}{2} A\zeta_2(\eta)\varepsilon_0^2 + I \left(\frac{1}{2}\zeta_1(t) + \eta\varepsilon_0 \right) \varepsilon_1^2 \\ + \frac{GA}{2E_b} \gamma_0^2 (k_1 + k_2\gamma_0^2), \end{aligned} \quad (\text{B.4a})$$

where the coefficients $\zeta_1(t)$ and $\zeta_2(\eta)$ are given by

$$\zeta_1(t) = 1 + 2 \left(\frac{\bar{C}_2 \bar{C}_B}{\bar{C}_1^2} + \frac{3}{20} \frac{\bar{C}_3}{\bar{C}_1} \right) t^2 + \frac{3}{112} \left(\frac{\bar{C}_3}{\bar{C}_1} \right)^2 t^4, \quad (\text{B.4b})$$

$$\zeta_2(\eta) = 1 + \frac{1}{6} \frac{\bar{C}_2}{\bar{C}_0} \left(1 + \frac{3}{40} \frac{\bar{C}_2}{\bar{C}_0} + \frac{3}{20} \eta \frac{\bar{C}_1^2}{\bar{C}_0} \right), \quad (\text{B.4c})$$

and where k_1 and k_2 are given by

$$k_1 = 1 + \frac{1}{6} \frac{\bar{J}_2}{\bar{J}_0} + \frac{1}{80} \left(\frac{\bar{J}_2}{\bar{J}_0} \right)^2, \quad (\text{B.4d})$$

and

$$k_2(t) = \frac{1}{12} \frac{\bar{J}_1^2}{\bar{J}_0^4} t^{-4}. \quad (\text{B.4e})$$

456 To obtain the above results we have used the fact that there is a clear pattern in
 457 the scaling exponents of the higher order corrections of the uniaxial and shear
 458 strain profiles with the excess displacement Δu , which alternate between 1/2 or
 459 1 (see Tables A.1 and A.3). Consequently, we can factorize the Δu dependence
 460 and express the higher order corrections in terms of the Mindlin-Reissner strains.
 461 For example, the quadratic postbuckling profile coefficient of the axial strain
 462 profile, $C_2 = \bar{C}_2 \Delta u t^4$, can be expressed in terms of $\varepsilon_0 \equiv C_0 = \bar{C}_0 \Delta u t^4$ as
 463 $C_2 = (\bar{C}_2/\bar{C}_0) \varepsilon_0$.

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